

CHAPTER 4 — AIRWORTHINESS LIMITATIONS SCHEDULE

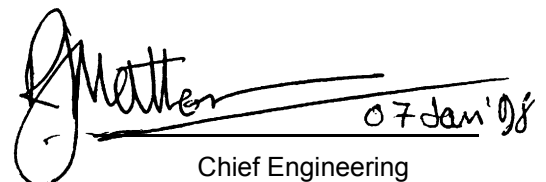
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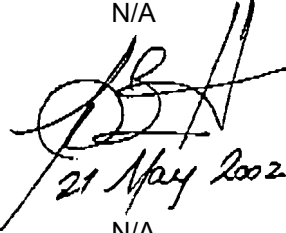
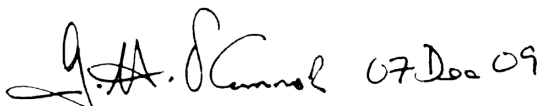

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The Airworthiness Limitations Schedule is approved by the Minister and specifies the maintenance required by any applicable airworthiness or operational rules unless an alternative program has been approved by the Minister.



07 Jan '08

Chief Engineering
Aircraft Certification
Transport Canada

| REVISION NO. | DATE OF SIGNATURE | TC SIGNATURE |
|--------------|-------------------|--|
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AIRWORTHINESS LIMITATIONS SCHEDULE

4-1. AIRWORTHINESS LIMITATIONS SCHEDULE

WARNING

ALL REPAIR AND OVERHAUL PROCEDURES LIVES PUBLISHED BY BELL HELICOPTER TEXTRON, INCLUDING COMPONENT RETIREMENT LIFE, ARE BASED SOLELY ON THE USE OF BELL APPROVED PARTS AND PROCESSES. IF PARTS OR PROCESSES DEVELOPED OR APPROVED BY PARTIES OTHER THAN BELL HELICOPTER ARE USED, THEN THE DATA PUBLISHED OR OTHERWISE SUPPLIED BY BELL HELICOPTER ARE NOT APPLICABLE. THE USER IS WARNED TO NOT RELY ON BELL HELICOPTER DATA FOR PARTS AND PROCESSES NOT APPROVED BY BELL HELICOPTER. ALL APPLICABLE INSPECTIONS AND REPAIR METHODS MUST BE OBTAINED FROM THE SUPPLIER OF THE PARTS OR PROCESSES NOT APPROVED BY BELL HELICOPTER. BELL HELICOPTER IS NOT RESPONSIBLE FOR PARTS OR PROCESSES OTHER THAN THOSE WHICH IT HAS ITSELF DEVELOPED OR APPROVED.

WARNING

CONTINUED USE OF ANY TIME/CALENDAR/RIN LIFE LIMITED COMPONENT PAST ITS STATED LIMITS IS NOT PERMITTED AS IT COULD SERIOUSLY AFFECT THE AIRWORTHINESS OF THE HELICOPTER.

THE COMPONENT MUST BE REMOVED FROM THE HELICOPTER NOT LATER THAN THE END OF THE LIFE LIMIT. IT MUST EITHER BE MADE UNSERVICEABLE OR MARKED IN A CONSPICUOUS MANNER TO PREVENT CONTINUED USE, EVEN INADVERTENT USE.

THE TIME/CALENDAR/RIN LIFE MUST NOT BE EXCEEDED WITHOUT THE EXPRESS WRITTEN AGREEMENT OF BELL HELICOPTER AND THE LOCAL CERTIFICATION AUTHORITIES.

WHEN A REPLACEMENT PART IS INSTALLED, THE PART NAME, PART NUMBER, SERIAL NUMBER, AND CURRENT OPERATING HOURS, IF APPLICABLE, MUST BE RECORDED IN THE HISTORICAL SERVICE RECORD FOR THE ASSEMBLY OR HELICOPTER.

NOTE

The requirements stated in [Chapter 5](#) are to be complied with and the appropriate maintenance actions are to be performed.

NOTE

The airworthiness life or inspection interval for any part number contained in this schedule applies to all the successive dash numbers for that component unless it is otherwise specified.

The mandatory airworthiness limitations schedule ([Table 4-1](#)) summarizes the mandatory maximum life, in hours, years (months) or by Retirement Index Number (RIN) of components with a limited airworthiness life. Parts that are not on the schedule have an unlimited airworthiness life.

Refer to the engine manufacturer's publications for the airworthiness limitations schedule of the engine and components.

WARNING

SOME PARTS ARE INSTALLED AS ORIGINAL EQUIPMENT ON BOTH MILITARY AND COMMERCIAL HELICOPTERS AND MAY HAVE A LOWER AIRWORTHINESS LIFE AND/OR OVERHAUL SCHEDULE WHEN USED ON A MILITARY HELICOPTER. IN ADDITION, CIRCUMSTANCES SURROUNDING THEIR USE MAY CALL FOR OPERATION OF THE MILITARY HELICOPTER OUTSIDE OF THE APPROVED COMMERCIAL FLIGHT ENVELOPE. CONSEQUENTLY, PARTS THAT HAVE BEEN USED ON MILITARY HELICOPTERS SHOULD NOT BE USED ON COMMERCIAL HELICOPTERS.

CAUTION

AIRWORTHINESS LIFE OF SOME KIT COMPONENTS MAY NOT BE COVERED IN THIS SCHEDULE. REFER TO APPLICABLE SERVICE INSTRUCTION

(SI) OR MAINTENANCE MANUAL SUPPLEMENT (MMS) FOR KIT COMPONENTS' SCHEDULE.

NOTE

The airworthiness life given or the failure to give an airworthiness life to a component does not constitute a warranty of any kind. The only warranty applicable to the helicopter or any component is the warranty included in the Purchase Agreement for the helicopter or the component.

The airworthiness lives given to the components and assemblies are determined by experience, tests and the judgment of Bell Helicopter engineers. The airworthiness lives and inspection intervals cannot be changed without the approval of the Minister of Transport Canada.

Prior to disposing of unsalvageable helicopter parts and materials, caution should be exercised to ensure that the parts and materials are disposed of in a manner that does not allow them to be returned for service. Refer to FAA Advisory Circular 21-38 for guidance on the disposal of unsalvageable helicopter parts and materials.

Table 4-1. Airworthiness Limitations Schedule








| COMPONENT | PART NUMBER  | AIRWORTHINESS LIFE |
|----------------------------------|---|--|
| MAIN ROTOR HUB AND BLADES | | |
| Yoke | 206-010-101-001/-009/-013/ -019/-021 |  10 years |
| Grip | 206-010-102-009/-013/-121 | 4800 hours |
| Trunnion | 206-010-104-003 |  2400 hours |
| Retention Strap | 206-010-105-003/-005 |  |
| Retention Strap Pin | 206-010-123-001 |  100 hours |
| Retention Strap Pin | 206-010-123-003 |  1200 hours |
| Retention Strap Fitting | 206-010-155-007 |  100 hours |

Table 4-1. Airworthiness Limitations Schedule (Cont)





















| COMPONENT | PART NUMBER  | AIRWORTHINESS LIFE |
|---|---|--|
| MAIN ROTOR HUB AND BLADES (CONT) | | |
| Retention Strap Fitting | 206-010-155-011/-015 |  1200 hours |
| Latch Bolt | 206-010-169-001/-003 |  |
| Blade | 206-010-200-029 |  |
| Blade | 206-010-200-033/-133/-137/ -139/-141 |  5000 hours |
| Trunnion | 206-011-113-001 |  2400 hours |
| Trunnion | 206-011-113-103 |  4800 hours |
| Latch Bolt | 206-011-122-003 |  |
| Retention Strap Fitting | 206-011-124-001 | 2400 hours |
| Retention Strap Pin | 206-011-125-001 | 1200 hours |
| Retention Strap Pin | 206-011-125-105 | 2500 hours |
| Grip | 206-011-132-009/-113 |  4800 hours |
| Retention Strap Fitting | 206-011-140-001 | 1200 hours |
| Retention Strap | 206-011-147-001/-003/-005/-007 |   1200 hours/24 months |
| Retention Strap Fitting | 206-011-150-101/-105 | 2400 hours |
| Retention Strap | 206-011-154-101/-103/-105/-107 |   1200 hours/24 months |
| Latch Bolt | 206-011-260-101 |   1200 hours |
| Latch Bolt | 206-011-260-103 |  2500 hours |
| SWASHPLATE AND SUPPORT | | |
| Upper Collective Tube | 206-001-024-017/-027 |  4800 hours |
| Lower Collective Tube | 206-001-194-001 | 4800 hours |
| Lower Cyclic Tube | 206-001-194-005 | Unlimited |
| Lower Cyclic Tube | 206-001-194-101 |  15,000 hours |
| Bellcrank Support | 206-001-521-005/-006 |  |
| Bellcrank Support | 206-001-521-013/-014 |  |
| Lower Collective Tube | 206-001-524-001/-011/-015 | 4800 hours |

Table 4-1. Airworthiness Limitations Schedule (Cont)















| COMPONENT | PART NUMBER  | AIRWORTHINESS LIFE |
|--------------------------------------|---|--|
| SWASHPLATE AND SUPPORT (CONT) | | |
| Servo Actuator Support | 206-001-566-001/-101 | 10,000 hours |
| Servo Actuator Support | 206-001-566-105 |  5000 hours |
| Collective Idler Link | 206-010-407-001 | 4800 hours |
| Swashplate Duplex Bearing | 206-010-420-001/-003 | 1200 hours |
| Swashplate Support Assembly | 206-010-452-001/-005/-109 |  4800 hours |
| Swashplate Support Assembly | 206-010-452-113 | 4800 hours |
| Swashplate Sleeve Assembly | 206-010-454-001/-005 | 4800 hours |
| Swashplate Sleeve Assembly | 206-010-454-109/-113 | 14,400 hours |
| Collective Lever | 206-010-467-001/-105 | 4800 hours |
| Servo Actuator | 206-076-023-001/-003 |  875 hours |
| Servo Actuator | 206-076-024-ALL |  875 hours |
| Servo Actuator | 41103750/-2 |  875 hours |
| Servo Actuator | 41103650-005 |  875 hours |
| POWER TRAIN | | |
| Mast | 206-010-332-001 |  1200 hours |
| Spider | 206-040-041-001 |  500 hours |
| Tail Rotor Gearbox Duplex Bearing | 206-040-410-003/-005/-101 |  3000 hours |
| Sun Gear | 206-040-662-101 |  |
| Oil Cooler Fan Impeller | 206-061-432-011 |  |
| Freewheel Clutch | CL41742/-1 | 3000 hours |
| Freewheel Clutch | CL42250-1 | 3000 hours |
| Freewheel Clutch | X136028 | 1200 hours |
| Oil Cooler Fan Impeller | SPL610-216 | 1200 hours |
| TAIL ROTOR HUB AND BLADES | | |
| Tail Rotor Blade | 206-010-750-003/-005/-007 |  |
| Tail Rotor Trunnion | 206-010-776-001 |  2400 hours |

Table 4-1. Airworthiness Limitations Schedule (Cont)































| COMPONENT | PART NUMBER  | AIRWORTHINESS LIFE |
|--|---|--|
| TAIL ROTOR HUB AND BLADES (CONT) | | |
| Tail Rotor Yoke | 206-010-788-001 |  1200 hours |
| Tail Rotor Yoke | 206-011-802-001 |  1200 hours |
| Tail Rotor Trunnion | 206-011-803-001/-005 |  2400 hours |
| Tail Rotor Yoke | 206-011-811-001/-005 |  |
| Tail Rotor Yoke | 206-011-811-009 |  4800 hours |
| Tail Rotor Yoke | 206-011-811-123/-125 |  1800 hours |
| Tail Rotor Yoke | 206-011-819-101/-105/-109 |  5000 hours |
| Tail Rotor Blade | 206-016-201-001/-103/-107/ -113/-115 |  2400 hours |
| Tail Rotor Blade | 206-016-201-125/-127/-131/ -133/-135/-137 |   2500 hours |
| Tail Rotor Blade | 206-016-501-001 | 200 hours |
| MISCELLANEOUS | | |
| Landing Gear Crosstube Assembly | 206-050-107-ALL |  21 |
| Emergency Lightweight Flotation Landing Gear Crosstube Assembly | 206-050-107-025/-027/-119/-121 |  37 |
| Landing Gear Crosstube Assembly | 206-050-119-ALL |  21 |
| Emergency Flotation Landing Gear Crosstube Assembly | 206-050-134-001/-003/-005/ -007/-011/-105 |  36 |
| Fixed Float Landing Gear Crosstube Assembly | 206-050-157-001/-003 |  38 |
| Fixed Float Landing Gear Crosstube Assembly | 206-050-169-001/-003 |  35 |
| Emergency Floats Cylinder | 206-073-818-001 |  10 years |
| Emergency Floats Cylinder | 206-073-839-103 |  15 years |
| Emergency Floats Cylinder | 206-073-848-113/ -121 (1271226) |  15 years |

Table 4-1. Airworthiness Limitations Schedule (Cont)

| COMPONENT | PART NUMBER  | AIRWORTHINESS LIFE |
|-----------------------------|---|--|
| MISCELLANEOUS (CONT) | | |
| Emergency Floats Cylinder | 206-073-919-001/-003/-005/-007 |  10 years |
| Emergency Floats Cylinder | 206-073-922-101 |  10 years |
| Emergency Floats Cylinder | 206-073-929-003/-101/-103 |  10 years |
| Emergency Floats Cylinder | 206-073-930-101 |  10 years |
| Emergency Floats Cylinder | 1269556-3/-4 |  15 years |
| Emergency Floats Cylinder | 23711982 |  15 years |
| Emergency Floats Cylinder | 274645 |  15 years |
| Squib, Emergency Floats | 94455 |  15 |
| POWER PLANT | | |
| Turboshaft Engine | 250-C18 Series | Refer to Rolls-Royce 250-C18 Series Operation and Maintenance Manual, 5W2 |
| Turboshaft Engine | 250-C20 Series | Refer to Rolls-Royce 250-C20 Series Operation and Maintenance Manual, 10W2 |
| AIRFRAME | | |
| Pylon Support Link | 206-031-508-005/-007 |  13 |

NOTES:






-  Airworthiness limitation for part number listed applies to all successive dash numbers for that component unless otherwise specified.
-  Yokes 206-010-101-125/-129 are not listed and are conditional. All yokes listed have a 10-year airworthiness life. Yokes in use as of 17 July 1979, with less than 5 years in service, shall be removed after 10 years. Yokes in use as of 17 July 1979, with more than 5 years in service, will have been removed by 1 January 1984. Refer to [ASB 206-79-4](#), dated 17 July 1979.
-  Trunnions 206-010-104 and 206-011-113-001 with more than 2400 hours in service, or time in use unknown, will have been removed from service by 1 September 1981. Refer to [ASB 206-80-7](#) Rev. B, dated 15 October 1980.
-  Retention strap fittings 206-010-155-011/-015 will have been removed from service no later than 1 January 1979. Refer to [ASB 206-76-7](#), dated 13 August 1976 and note .

Table 4-1. Airworthiness Limitations Schedule (Cont)
NOTES (CONT):

△₅ Grip 206-011-132-009, with serial numbers listed below, have a 1200-hour airworthiness life. The serial numbers not listed below have a 4800-hour airworthiness life. These parts are originally delivered on Bell 206L1, but could have found their way to the 206A/B series. Refer to [ASB 206L-80-15](#), dated 29 July 1980.

| | | | | | |
|----------|----------|----------|----------|----------|----------|
| A1-00001 | A1-00029 | A1-00050 | A1-00077 | A1-00099 | A1-00132 |
| A1-00004 | A1-00031 | A1-00053 | A1-00078 | A1-00100 | A1-00133 |
| A1-00005 | A1-00032 | A1-00054 | A1-00080 | A1-00101 | A1-00134 |
| A1-00006 | A1-00033 | A1-00055 | A1-00083 | A1-00103 | A1-00135 |
| A1-00007 | A1-00034 | A1-00056 | A1-00084 | A1-00104 | A1-00136 |
| A1-00009 | A1-00035 | A1-00057 | A1-00085 | A1-00105 | A1-00192 |
| A1-00010 | A1-00036 | A1-00059 | A1-00086 | A1-00106 | A1-00222 |
| A1-00014 | A1-00038 | A1-00061 | A1-00087 | A1-00111 | A1-00248 |
| A1-00016 | A1-00039 | A1-00062 | A1-00088 | A1-00113 | A1-00287 |
| A1-00017 | A1-00040 | A1-00064 | A1-00089 | A1-00115 | A1-00300 |
| A1-00019 | A1-00042 | A1-00066 | A1-00090 | A1-00116 | A1-00305 |
| A1-00021 | A1-00043 | A1-00067 | A1-00091 | A1-00118 | A1-00308 |
| A1-00022 | A1-00044 | A1-00068 | A1-00092 | A1-00119 | |
| A1-00023 | A1-00046 | A1-00071 | A1-00093 | A1-00120 | |
| A1-00024 | A1-00047 | A1-00073 | A1-00095 | A1-00124 | |
| A1-00027 | A1-00048 | A1-00074 | A1-00096 | A1-00130 | |
| A1-00028 | A1-00049 | A1-00075 | A1-00098 | A1-00131 | |

△₆ Retention straps 206-010-105-003/-005 will have been removed from service no later than 1 January 1979. Refer to [ASB 206-78-1](#), dated 14 April 1978 and FAA AD 78-11-02 R1, dated 15 December 1988.

△₇ Retention straps 206-011-147 and 206-011-154 have airworthiness lives of 1200 hours or 24 months, whichever occurs first. The calendar life of 24 months starts when new straps are installed in a main rotor hub and blade assembly, and are subjected to rotation on the helicopter. Refer to [ASB 206-80-9](#), dated 3 June 1980.

△₈ Main rotor blade 206-010-200-029 will have been retired no later than 31 March 1976. Refer to [ASB 206-75-10](#), dated 13 November 1975.

△₉ Swashplate supports 206-010-452-001/-005/-109 with more than 1500 hours in service are subjected to a visual inspection every 50 hours. The supports may remain in service until a crack is found or until a total of 4800 hours time in service is accumulated. Refer to [ASB 206-93-74](#) Rev. B, dated 4 April 1994.

Table 4-1. Airworthiness Limitations Schedule (Cont)
NOTES (CONT):

- ▲₁₀ Tail rotor blade 206-010-750 will have been retired no later than 1 July 1981. Refer to [ASB 206-80-12](#), dated 9 September 1980.
- ▲₁₁ Tail rotor yokes 206-010-788, 206-011-802 and trunnions 206-010-776-001, 206-011-803-001 are not authorized after 1 July 1981. Yokes are to be replaced concurrently with the replacement of tail rotor blade 206-010-750. See also note ▲₁₀. Refer to [ASB 206-75-4](#), dated 15 May 1975.
- ▲₁₂ Tail rotor yokes 206-011-811-001/-005 are not authorized after 1 July 1981. Yokes were reidentified as 206-011-811-123/-125 concurrently with the replacement of tail rotor blade 206-010-750. Yokes that have been reidentified have an 1800-hour airworthiness life starting with the reidentification. Refer also to note ▲₁₀.
- ▲₁₃ Aluminum links 206-031-508-005/-007 have been removed from service. Refer to [ASB 206-01-73-6](#), dated 19 September 1973 and FAA AD 73-8-3, dated 18 April 1973.
- ▲₁₄ Duplex bearings 206-040-410-001/-103 installed in tail rotor gearbox 206-040-400-(ALL) are conditional. Duplex bearings 206-040-410-003/-005/-101 installed in tail rotor gearbox 206-040-402-(ALL) have a life of 3000 hours.
- ▲₁₅ Cartridge 94455 has a 5-year shelf life and a 5-year in-service life. These lives can be combined for a total of 10 years. Refer to [IL GEN-93-52](#), dated 21 June 1993.
- ▲₁₆ Selected serial numbers of main rotor trunnion 206-011-113-103 will have been removed from service no later than 30 November 1994. Refer to [ASB 206-94-80](#), dated 5 August 1994 for list of serial numbers affected.
- ▲₁₇ Latch bolts 206-011-260-101/-103 that have a serial number with the prefix "SC", purchased as spares will have been removed from service no later than 3 December 1992. Refer to [ASB 206-92-70](#), dated 3 December 1992.
- ▲₁₈ Selected serial numbers of tail rotor hub assemblies 206-011-810-015 and tail rotor yokes 206-011-811-009 must be removed from service. Refer to [ASB 206-89-47](#) Rev. A, dated 29 August 1989 for list of serial numbers affected.
- ▲₁₉ Retention strap pin 206-010-123-001 and fitting 206-010-155-007 have been removed from service. Refer to FAA AD 72-19-1, dated 7 September 1972.
- ▲₂₀ Refer to [ASB 206\(04-1\)-73-1](#), dated 25 June 1973.
- ▲₂₁ Crosstube assemblies 206-050-107-(ALL) and 206-050-119-(ALL) must be removed from service. Refer to [ASB 206-94-78](#), dated 11 April 1994.
- ▲₂₂ Usable only with main rotor hub 206-011-100-127. Refer to [TB 206-91-133](#) Rev. A, dated 1 July 1998.
- ▲₂₃ Usable only on model 206B, S/N 5101 and subsequent. These part numbers are reserved for US Army, model TH-67 training helicopters. Refer to [ASB 206-96-87](#) Rev. A, dated 22 March 1996.
- ▲₂₄ Oil cooler impeller 206-061-432-011 will have been removed from service no later than 1 March 1978. Refer to [ASB 206-77-10](#) Rev. A, dated 21 February 1978.

Table 4-1. Airworthiness Limitations Schedule (Cont)**NOTES (CONT):**




















-  Selected serial numbers of tail rotor yokes 206-011-819-101 must be removed from service. Refer to [ASB 206-88-41](#), dated 28 January 1988 for list of serial numbers affected.
-  Selected serial numbers of retention straps 206-011-147-007 and 206-011-154-101/-105 must be removed from service. Refer to [ASB 206-89-46](#) Rev. A, dated 29 August 1989 for list of serial numbers affected.
-  Selected serial numbers of main rotor blades 206-010-200-033 must be removed from service. Refer to [ASB 206-89-50](#) Rev. A, dated 29 November 1989 for list of serial numbers affected.
-  Selected serial numbers of tail rotor blades 206-016-201-113/-127 must be removed from service. Refer to [ASB 206-90-57](#), dated 27 June 1990 for list of serial numbers affected.
-  Selected serial numbers of tail rotor blades 206-016-201-125/-127 must be removed from service. Refer to [ASB 206-90-58](#) Rev. A, dated 31 October 1990 for list of serial numbers affected.
-  Refer to [ASB 206A-12](#) Rev. A, dated 8 April 1970. Servo actuator 206-076-023 is equivalent to Hydraulic Research part number 41103750 and servo actuator 206-076-024 is equivalent to Hydraulic Research part number 41103650.
-  Refer to [ASB 206A-14](#) Rev. A, dated 8 April 1970.
-  Cyclic control bellcrank support assembly 206-001-521-005/-006 will have been removed from service no later than 15 July 1969. Refer to [ASB 206A-11](#) Rev. B, dated 29 September 1969.
-  Masts 206-010-332-005/-009/-013/-017/-121 are not listed and are conditional. Selected serial numbers of main rotor mast 206-010-332-121 will have been returned to Bell Helicopter Textron for metallurgical evaluation no later than 31 May 1987. Affected masts inspected at BHT and found to be acceptable have a suffix "U" added to the serial number. Refer to [ASB 206-87-37](#), dated 23 April 1987 for list of serial numbers affected.
-  Selected serial numbers of sun gear 206-040-662-101 will have been removed from service no later than 15 January 1991. Refer to [ASB 206-90-56](#) Rev. A, dated 15 January 1991 for list of serial numbers affected.
-  Fixed float landing gear crosstubes 206-050-169-001/-003 must be removed from service. Refer to [ASB 206-94-81](#) Rev. A, dated 17 August 1994.
-  Emergency flotation landing gear crosstubes 206-050-134-001/-003/-005/-007/-011/-103/-105 must be removed from service. Refer to [ASB 206-94-82](#) Rev. A, dated 17 April 1995.
-  Emergency lightweight flotation landing gear crosstubes 206-050-107-025/-027/-119/-121 must be removed from service. Refer to [ASB 206-94-83](#), dated 6 September 1994.
-  Fixed float landing gear crosstubes 206-050-157-001/-003 must be removed from service. Refer to [ASB 206-96-88](#), dated 3 June 1996.
-  Usable only on model 206A, S/N 5001 through 5040. These part numbers are reserved for US Army, model TH-57 training helicopters.
-  Cylinder per DOT-3HT and DOT-E7218 specifications. Refer to [IL GEN-93-52](#), dated 21 June 1993.

Table 4-1. Airworthiness Limitations Schedule (Cont)

NOTES (CONT):

-  Cylinder per FAA specifications. Refer to [IL GEN-93-52](#), dated 21 June 1993.
-  Upper collective tube 206-001-024-029 is not listed and is conditional.
-  Main rotor latch bolts 206-010-169-001/-003 and 206-011-122-003 must be removed from service. Refer to [ASB 206-06-109](#), dated 25 July 2006, and Transport Canada AD [CF-2006-23](#), dated 6 November 2006.
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